

Birla Institute of Technology and Science, Pilani
ME F485 – Numerical Methods in Fluid Flow and Heat Transfer
Mid-Semester Examination, Spring 2017-2018
(Closed Book)

Date:10/3/2018

Duration: 9-10:30AM

Marks [75]

1. Multiple Choice Questions. (1M for each correct choice, -0.5 M for incorrect choice) [15M]

a. Select all that apply for TDMA.

- | | |
|------------------------------------------------------------|-----------------------|
| i. Tridiagonal Solver | v. LU decomposition |
| ii. Thomas Algorithm | vi. Gauss Elimination |
| iii. Indirect solution method | vii. Direct Solver |
| iv. $O(n)$ operations, where n is the size of the system | |

b. For a certain 2D elliptic problem, Jacobi scheme leads to non-convergence. What options are at disposal to a numerical scientist to overcome convergence issues?

- | | |
|----------------------------------|--------------------------------------------|
| i. Gauss-Seidel iterative scheme | iv. Higher order finite difference schemes |
| ii. Increase mesh refinement | v. Alternate Direct Implicit |
| iii. Under-relaxation | vi. Employ Direct Solvers |

c. Select the statements that apply to Multigrid Framework

- | | |
|-------------------------------------------------------|--------------------------------------------------------------------------------------|
| i. More than one mesh refinements used | v. Achieves speed up by operating on coarser grids where spectral radius is smaller. |
| ii. Offers good compromise between speed and accuracy | vi. Offers accuracy of finest grid in work units similar to the coarsest grid. |
| iii. Free of truncation errors | |
| iv. Prolongation is akin to interpolation operation | |

d. Select statements that pertain to hyperbolic equation,

- | | |
|------------------------------------------------------------------------------|-------------------------------------------------------------------------------------------------|
| i. Any point in solution space is only affected by its domain of disturbance | iii. Effect of disturbance travels at a finite wave speed along its characteristics directions. |
| ii. Solution is a hyperbolic function | iv. Unsteady heat conduction equation |
| | v. Must have discontinuity in the solution |

2. Classify the following equations into hyperbolic, elliptic, parabolic. [5M]

a. $\frac{\partial \phi}{\partial t} + c \frac{\partial \phi}{\partial x} = 0$

b. $\frac{\partial \phi}{\partial t} - \Gamma \frac{\partial^2 \phi}{\partial x^2} = 0$

c. $(1 - M^2) \frac{\partial^2 \phi}{\partial x^2} + \frac{\partial^2 \phi}{\partial y^2} = 0$

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3. Consider the following non-uniform grid. Derive a central finite difference scheme for $\left. \frac{\partial \phi}{\partial x} \right|_i$ as an average of forward difference and backward difference in terms of Δx_- and expansion ratio/growth ratio $g = \frac{\Delta x_+}{\Delta x_-}$. Show at least 2 terms in truncation error. What is the Leading Error Term(LET)? Is the scheme still second order? Comment on effect of g on truncation error.[15M]

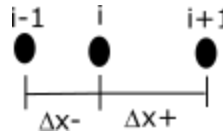
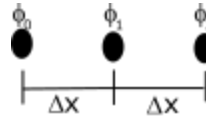


Figure Q3. Non-Uniform Finite Difference Grid

4. Consider integral form of governing equation over the domain Ω , $\int_{\Omega} \frac{d^2 \phi}{dx^2} d\Omega = \int_{\Omega} S d\Omega$. Approximate piecewise polynomial solution of the form $\hat{\phi}(x) = ax^2 + bx + c$ is sought. Devise second order forward Finite Difference scheme to express equation for left boundary node ϕ_0 (Figure 2) having Neumann boundary condition prescribed, $\left. \frac{d\phi}{dx} \right|_0 = f$. [15M]

Figure Q4. 1-Dimensional uniform finite difference grid



5. The gradient of pressure can be approximated by

$$(A) \left. \frac{\partial p}{\partial x} \right|_i = \frac{p_{i+1} - p_{i-1}}{2\Delta x} \quad \text{or} \quad (B) \left. \frac{\partial p}{\partial x} \right|_i = \frac{p_{i+1/2} - p_{i-1/2}}{\Delta x}$$

Which one do you think will be more accurate? Find the difference between the two in the form

$$(A) = (B) + a\Delta x^n \left. \frac{\partial^m p}{\partial x^m} \right|_i. \text{ Specify } a, n, \text{ and } m. \quad [15M]$$

$$\text{Taylor Series: } p(x + \Delta x) = p(x) + \Delta x p'(x) + \frac{\Delta x^2}{2} p''(x) + \frac{\Delta x^3}{6} p'''(x) + \frac{\Delta x^4}{24} p''''(x) + \dots$$

6. What simplification can you make to the following generalized conservation equations for incompressible, inviscid, steady air flow over an airfoil? Justify your assumptions. [10M]

Continuity:
$$\frac{\partial \rho}{\partial t} + \vec{\nabla} \cdot (\rho \vec{V}) = 0$$

Momentum Cons.:
$$\frac{\partial(\rho \vec{V})}{\partial t} + \vec{\nabla} \cdot (\rho \vec{V} \vec{V}) = -\vec{\nabla} p + \vec{\nabla} \cdot \vec{\tau} + \vec{B}$$

Energy Cons.:
$$\frac{\partial(\rho e)}{\partial t} + \vec{\nabla} \cdot (\rho e \vec{V}) = -\frac{\partial p}{\partial t} - \vec{\nabla} \cdot (p \vec{V}) + \vec{B} \cdot \vec{V} + \vec{\nabla} \cdot (\vec{\tau} \cdot \vec{V}) + \vec{\nabla} \cdot (k \vec{\nabla} T) + \dot{q}$$