

Birla Institute of Technology & Science, Pilani
First Semester 2023-2024
Comprehensive Exam

Course No. : ME G512
 Course Title : Finite Element Methods
 Nature of Exam : Open Book
 Weightage : 35%
 Duration : 3 hours
 Date of Exam : 09/12/2023

No. of Pages = 2
 No. of Questions = 6

Note to Students:

1. All parts of a question should be answered consecutively. Each answer should start from a fresh page.
2. Assumptions made if any, should be stated clearly at the beginning of your answer.

Q.1. The nodal displacement values (in mm) of the elements shown in **Figure Q1a** and **Figure Q1b** are $u_1 = 0.2645$, $u_2 = 0.2172$, $u_3 = 0.1800$ for the triangular element and $u_1 = 0.2173$, $u_3 = 0.1870$, $u_2 = u_4 = 0.2232$ for the rectangular element. Dimensions of the elements are in meters.

- a). Compute u , $\partial u/\partial x$ and $\partial u/\partial y$ at the point $(x, y) = (0.375 \text{ m}, 0.375 \text{ m})$ for *triangular element*.
- b). Compute u , $\partial u/\partial x$ and $\partial u/\partial y$ at the point $(x, y) = (0.375 \text{ m}, 0.375 \text{ m})$ for *rectangular element*.

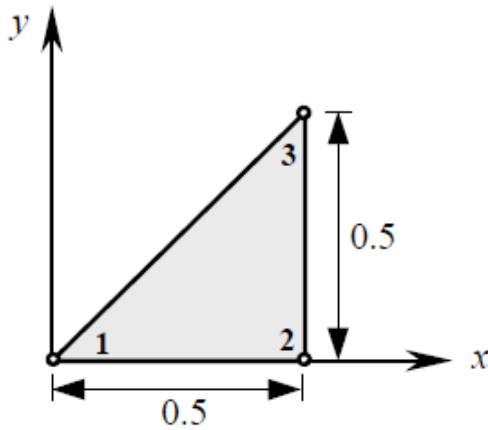


Figure Q1a

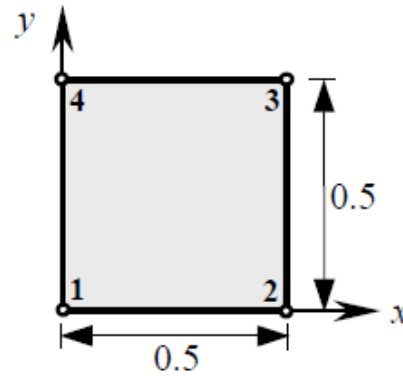


Figure Q1b

[3 + 3 = 6]

Q.2. (a) Determine the interpolation function ψ_9 in terms of area coordinates, L_i , for the triangular element shown in **Figure Q2a**.

(b) Consider the five-noded element shown in **Figure Q2b** and determine interpolation functions for the element in terms of natural coordinates, ξ and η .

[3 + 4 = 7].

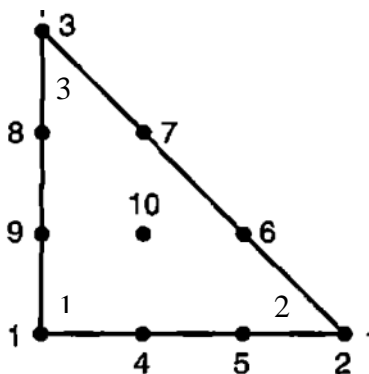


Figure Q2a

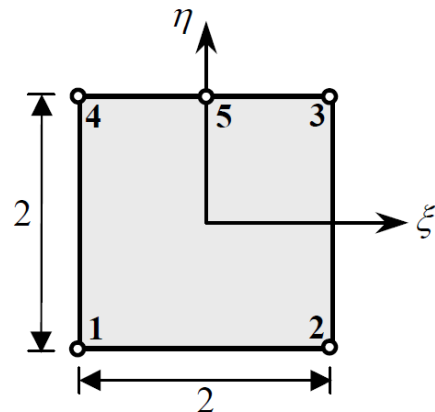


Figure Q2b

- Q.3. A bar member ($E = 200 \text{ GPa}$, $A = 40 \text{ cm}^2$) of an aerospace structure is discretised using **one 3-noded bar element**, as shown in the **Figure 3b** below. Determine the following:
- Interpolation functions for the 3-noded bar element shown in **Figure Q3b**.
 - Global Stiffness Matrix for the bar

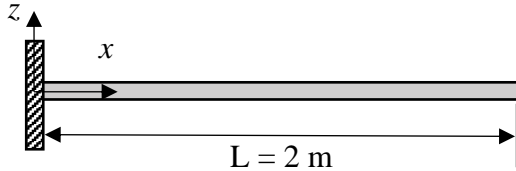


Figure Q3a

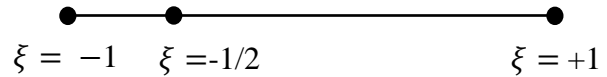


Figure Q3b

[3 + 4 = 7]

- Q.4. Determine the following for the beam shown in the **Figure Q4** using **two 2-noded Euler-Bernoulli's beam elements**.
- Global Stiffness Matrix
 - Deflection and Slopes at all nodes

[3 + 3 = 6]

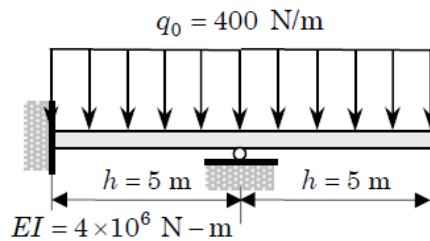


Figure Q4

- Q.5. A wall with thickness L , is maintained at temperature T_0 at the left end and is insulated at the right end. The thermal conductivity of the wall material varies linearly as follows:

$$k = k_0 \left(1 - \frac{x}{2L}\right) \text{ for } 0 < x < L, \text{ where } k_0 \text{ is a constant.}$$

Assume one-dimensional heat conduction through the wall, discretise the wall length using only one 2-noded isoparametric element and derive the element temperature matrix (K_{Te}).

[5]

- Q.6. The governing equation and boundary conditions for a structural phenomenon are given by:

$$\frac{d^2}{dx^2} \left(k_1 \frac{d^2 v}{dx^2} \right) + \lambda \frac{d^2 v}{dx^2} = 0 \quad \text{for } 0 < x < 1$$

$$v(0) = v(1) = 0, \quad \left(k_1 \frac{d^2 v}{dx^2} \right)_{\{x=0\}} = \left(k_1 \frac{d^2 v}{dx^2} \right)_{\{x=1\}} = 0$$

Where k_1 and λ are constants.

Derive the weak form of the governing equation and identify primary and secondary variables

[4]